# **Introduction**

### **Centroid / Centre of area**

The point where the area of a figure is assumed to be concentrated.

Located on the axes of symmetry.

### **First moment of area**

Measure of spatial distribution of a shape in relation to an axis.



Here:

- $\cdot$   $\bar{x}$  Centroid's  $x$  coordinate
- $\cdot$   $\bar{y}$  Centroid's  $y$  coordinate
- $\bullet$   $A$  Total area

About an axis of symmetry, first moment of area is  $0$ .

# **Second moment of area**

$$
\text{About x-axis} = I_{xx} = I_x = \int_A y^2 \text{ d}A
$$

About y-axis = 
$$
I_{yy} = I_y = \int_A x^2 dA
$$

Always positive.

#### **For common shapes**



**The product of moment of area about x,y axes**

$$
I_{xy}=\int_A xy\;\text{d}A
$$

**The polar moment of area about z axis**

$$
I_{zz}=J_0=\int_A r^2 \;\mathrm{d}A=I_{xx}+I_{yy}
$$

**Radius of gyration**

$$
\text{About x-axis} = r_x^2 = \frac{I_{xx}}{A}
$$

$$
\text{About y-axis} = r_y^2 = \frac{I_{yy}}{A}
$$

$$
\text{About z-axis} = r_z^2 = \frac{I_{zz}}{A}
$$

## **Parallel Axis Theorem**

$$
I_x = I_{x_1} + A\bar{y}^2
$$
  

$$
I_y = I_{y_1} + A\bar{x}^2
$$
  

$$
I_{xy} = I_{x_1y_1} + A\bar{x}\bar{y}
$$

#### **Here**

- On LHS, the moments of area are about some  $x$ ,  $y$  axes.
- On RHS, the moments of area are about centroidal axes  $x_1$ ,  $y_1$  parallel to x, y.
- $\bar{x}$  is the distance between  $x$  and  $x_2$  axes.
- $\bar{y}$  is the distance between  $y$  and  $y_1$  axes.

#### **Note**

 $\boldsymbol{I_x}$  is at a minimum when the axis is through the centroid. Same for  $\boldsymbol{I_y}.$ 

# **Perpendicular Axis Theorem**

 $I_{zz} = I_{xx} + I_{yy} = I_{mm} + I_{nn}$ 

x, y, z are a set of axes.  $m$ ,  $n$ ,  $z$  are another set of axes.

If  $I_{xx}$  is at maximum,  $I_{yy}$  will be at minimum.

# **Transformation Law**

The 2 sets of axes must share the origin.

#### **Note**

Don't have to memorize this. Will be given on exams, if required.



## **Principal Axes**

The product of moment of area is  $0$  about principal axes.

$$
I_{xy}=0
$$

There will be 2 directions of principal axes which are perpendicular to each other.

#### **Note**

For a shape with more than 2 axis of symmetry, all axes through the centroid is a principal axis.

## **Principal second moments of area**

Second moments about the principal axes.

About principal axes second moments of area will be at minimum and maximum.

### **Centroidal principal axes**

Principal axes through the centroid.

**Note**

Any axis of symmetry is a centroidal principal axis.

## **Beams**

- $\log (L >> B, D)$
- axis of the beam is straight
- constant cross-section throughout its length

## **Classified by supporting conditions**

First 3 are the mandatory ones in s1.

u.d.l means uniformly distributed load.





## **At a section**



- $\bullet$   $\mathbf{P}$  Normal force / Axial force
- $S_y, S_y$  Shear forces along  $y$  and  $z$  axis
- $M_x$  Twisting moment / Torque
- $M_y, M_z$  Bending moments about  $y$  and  $z$  axis

### **Degress of freedom**

A plane member have 3 degrees of freedom. Any of the 3 can be restrained.

- Displacement in  $x$ -direction
- Displacement in  $y$ -direction
- Rotation about  $z$ -direction

### **SFD & BMD**

#### **Sign convention**

- Bending moment
	- Hogging (curves upwards in the middle) is **(+) ve**
	- Sagging (curves downwards in the middle) is **(-) ve**
- Shear force
	- Clockwise shear is **(+) ve**.
	- Counterclockwise shear is **(-) ve**.

### **Pure bending**

A member is in pure bending when shear force is  $0$  and bending moment is a constant.

### **Point of Contraflexure**

The point about which bending moment is  $0$ , and changes its sign through the point.

### **Distributed load, shear force & bending moment**

Suppose a beam is under a distributed load of  $w = f(x)$  per unit length.

$$
\frac{\mathrm{d} S}{\mathrm{d} x} = -w
$$

$$
\frac{\mathrm{d}M}{\mathrm{d}x}=-S\hspace{2mm}\wedge\hspace{2mm}\frac{\mathrm{d}^2M}{\mathrm{d}x^2}=w
$$

## **Deflection of a beam**

Suppose a simply supported beam is applied a load of  $W$  at mid-span.

$$
S_{\max} = \frac{WL}{4I} \quad \land \quad D_{\max} = \frac{WL^3}{48EI}
$$

Here:

- $S_{\text{max}}$  Maximum stress
- $D_{\text{max}}$  Deflection
- $\bullet$   $W$  Load
- $\bullet$   $L$  Span length
- $\bullet$   $E$  Young's modulus
- $\bullet$   $I$  Second moment of cross-sectional area

# **Principle of Superposition**

A beam with multiple loads can be split into multiple systems each with a single load. Reason for doing so is the ease of calculations.

Values will be the sum of each system's corresponding value.

# **Structural Elements**

3 types:

- $\bullet$  Bars (1D)
- Plates and Shells (2D)
- Blocks (3D)

Bar elements are only focused for s1.

## **Pin Joint**

Doesn't exert a moment. Free rotations are allowed. When only pin joints are used, bars will have only axial forces.

### **Bars**



### **Here**

- $\bullet$   $N$  Axial force
- $S_x, S_y$  Shear force
- $\bullet$   $M_x$

### **Types of bars**

### **Axially loaded**

Generally in trusses, **pin joints** are considered.

- Predominant tension Ties
- Predominant compression Struts

### **Flexural**

• Predominant bending - beams

### **Torsional**

• Predominant torque - shafts

# **Trusses**

Also known as Ties-Struts model.

## **Definition**

An assembly of members used to span long distances. Idealized as

- **•** Connected by **frictionless** pin joints at their ends
- Developing axial forces

## **Types**

2 types:

- Plane truss (2D)
- Space truss (3D)

A truss requires 3 external reactions for equilibrium.

Predominant force is axial force.

### **Advantages of truss**

- High span
- Material efficiency

## **Triangulation**

To create a truss:

- Start with a triangle  $(3 \text{ bars and } 3 \text{ joints})$
- $\bullet$  Add  $2$  more bars and  $1$  joint repeatedly

This type of truss is a **simple truss**.

### **Simple (Closed) Truss**

When a truss is only made of bars and joints.

## **Open Truss**

When a truss is pinned directly to a foundation. It has 1 member & 2 free joints less than a closed truss.

## **Stability of trusses**

When a truss is:

- unstable: it's called a mechanism
- stable: it's called a structure

#### **Stable truss**

When the shape cannot be altered, the structure is **internally stable**.

#### **Stable & determinate (simply stiff)**

**Determinate** means internal forces can be determined by laws of statics alone.

#### **Stable & indeterminate**

**Indeterminate** means laws of statics alone are not sufficient to determine forces; relative stiffness of members will influence the solution (Indeterminate trusses are safer).

#### **Unstable truss**

When the shape can be altered, the truss is called a mechanism.

### **Necessary condition for being simply stiff**

#### **Note**

These are necessary (but not sufficient) conditions.

#### Here:

- $\bullet$   $m$  Number of members (bars)
- $\bullet$   $j$  Number of joints

#### **For a 2D simple (closed) truss**

- $\bullet$   $m < 2j-3$  truss is unstable
- $\cdot$   $m=2j-3$  truss is determinate if stable
- $m > 2j-3$  truss is indeterminate if stable

#### **For a 2D open truss**

- $\bullet~~ m < 2j$  truss is unstable
- $\bullet$   $m=2j$  truss is determinate if stable
- $m > 2j$  truss is indeterminate if stable

**For a 3D simple (closed) truss**

$$
m=3j-6
$$

**For a 3D open truss**

 $m=3j$ 

# **Analysis of Trusses**

Deviations from the ideal in real trusses.

- Loads are not applied only at joints; hence there is bending in members
- Joints are not perfectly pinned, so moments can be developed at joints

### **Method of Joints**

#### **Principle**

Since the truss is in equilibrium, each pin joint must also be in equilibrium.

#### **Note**

2 equilibrium equations can be written at each joint – vertical & horizontal.

#### **Sign convention**

Forces acting on each joint is marked. Tensile forces are positive. Compressive forces are negative.

#### **Method**

- Find external reactions using equilibrium equations for the entire truss.
- Start with a joint with only 2 unknown joint forces.
- Mark the forces (consider all forces are tensile) acting on the joint.
- Find the unknown forces at the selected joint, using 2 equilibrium equations for the joint.
- Go to all other joints in turn and find forces in all the members.

#### **Special cases**





### **Method of Sections**

#### **Principle**

Since the truss is in equilibrium, each of its section must be in stable equilibrium.

#### **Method**

- Decide on which member's internal force must be calculated.
- Cut the truss through **3 or less** members including the target member.
- $\bullet$  Internal forces in cut members become external forces. Can be represented as tensile forces.
- Use equilibrium equations for RHS or LHS section to find the internal forces.

# **Beam Analogy (Approximate) method**

In this method, the internal forces are found assuming the elongated truss is a beam.

#### **For a simply supported beam**

- $\bullet$  Maximum bending moment is at mid-span:  $M_{\rm max} = \frac{w L^2}{8}$
- Maximum shear force is at the supports:  $\frac{wL}{2}$



Here:

- Chord members horizontal members
- Web members diagonal members
- $\bullet$   $d$  truss depth

In the truss,

• Bending moment is carried by chord members.

### $\text{Bending moment} = F_{\text{chord}} \times d$

Shear force is carried by vertical component of web member force





# **Indeterminate Trusses**

When a truss is indeterminate, one or more compatibility equations (related to truss deformation) must be used.